

Interpolation of parameterized reduced-order models

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Outline

- Parameterized unsteady PDE
- Database approach
- Interpolation of reduced-order bases
- Interpolation of reduced-order matrices
- Applications

Parametrized unsteady PDE

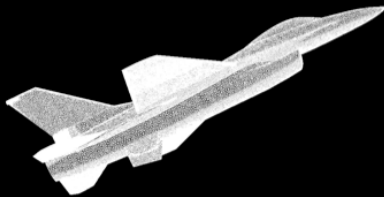
- Linear (linearized) PDE

$$\frac{d\mathbf{w}}{dt}(t) = \mathbf{A}(\boldsymbol{\mu})\mathbf{w}(t) + \mathbf{B}(\boldsymbol{\mu})\mathbf{u}(t)$$

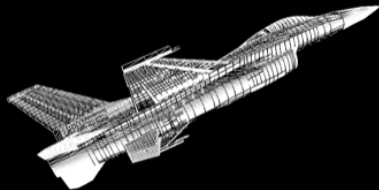
Example: aeroelasticity

- Example: aeroelastic analysis of full aircraft configuration

✦ F-16 Block 40



CFD model

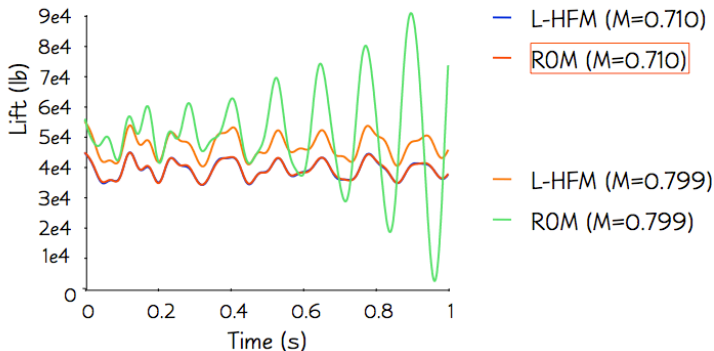


FEM structural model

- Hundreds of flight conditions $\mu = (M_\infty, \alpha)$ to clear for flutter

Example: aeroelasticity

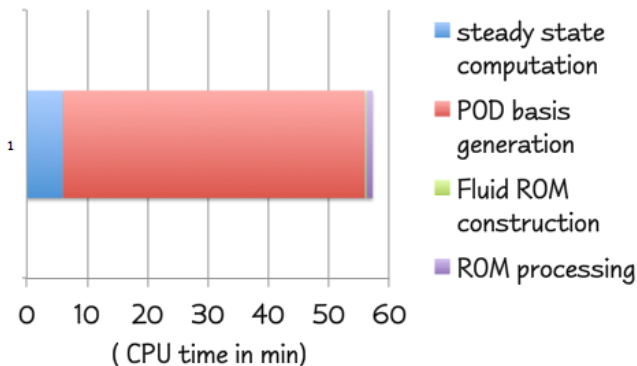
- Example



- Non-robustness with respect to the operating condition

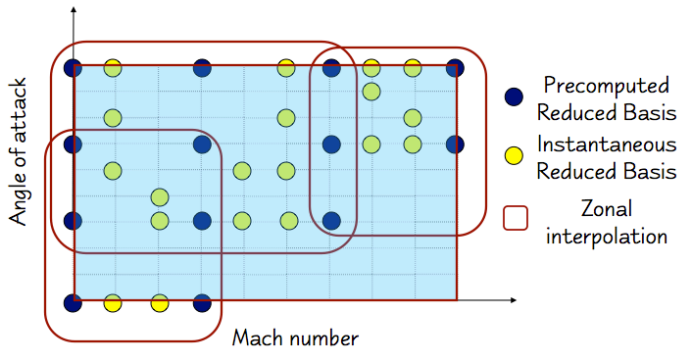
Example: aeroelasticity

- Can we afford rebuilding the POD basis?



- Not suitable for real-time analysis

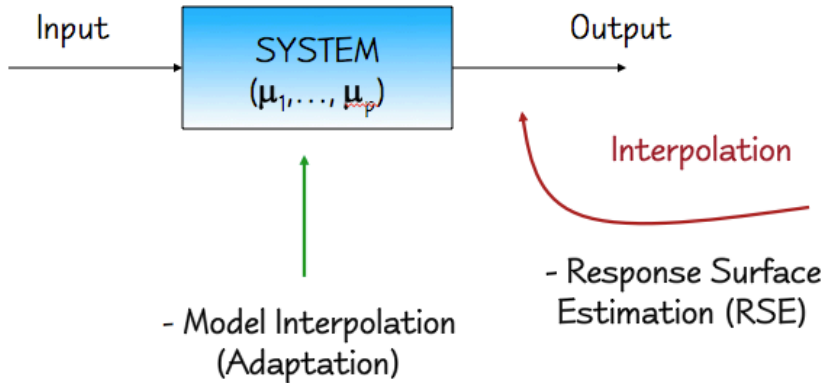
Database approach



- What should we interpolate?

Database approach

- What should we interpolate?



Model interpolation

- Reduced set of equations (after Galerkin projection)

$$\frac{d\mathbf{q}}{dt}(t) = \mathbf{V}(\boldsymbol{\mu})^T \mathbf{A}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}) \mathbf{q}(t) + \mathbf{V}(\boldsymbol{\mu})^T \mathbf{B} \mathbf{u}(t)$$

- Full state reconstruction

$$\mathbf{w}(t) \approx \mathbf{V}(\boldsymbol{\mu}) \mathbf{q}(t)$$

- Model: $(\mathbf{V}(\boldsymbol{\mu})^T \mathbf{A}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}), \mathbf{V}(\boldsymbol{\mu})^T \mathbf{B}, \mathbf{V}(\boldsymbol{\mu}))$
- Approach #1:
 - 1 interpolate $\mathbf{V}(\boldsymbol{\mu})$
 - 2 evaluate $(\mathbf{A}(\boldsymbol{\mu}), \mathbf{B}(\boldsymbol{\mu}))$
 - 3 form $(\mathbf{V}(\boldsymbol{\mu})^T \mathbf{A}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}), \mathbf{V}(\boldsymbol{\mu})^T \mathbf{B})$

Direct interpolation

- Natural idea: interpolate $\mathbf{V}(\boldsymbol{\mu}) \in \mathbb{R}^{N_{\mathbf{w}} \times k}$ entry-by-entry
- Input:
 - target $\boldsymbol{\mu}$
 - precomputed reduced bases $\{\mathbf{V}(\boldsymbol{\mu}_l)\}_{l=1}^m$
 - multi-variate interpolation operator $a(\boldsymbol{\mu}) = \mathcal{I}(\boldsymbol{\mu}; \{a(\boldsymbol{\mu}_l), \boldsymbol{\mu}_l\}_{l=1}^m)$
- Algorithm
 - 1: **for** $i = 1 : N_{\mathbf{w}}$ **do**
 - 2: **for** $j = 1 : k$ **do**
 - 3: Compute $v_{ij}(\boldsymbol{\mu}) = \mathcal{I}(\boldsymbol{\mu}; \{v_{ij}(\boldsymbol{\mu}_l), \boldsymbol{\mu}_l\}_{l=1}^m)$
 - 4: **end for**
 - 5: **end for**
 - 6: Form $\mathbf{V}(\boldsymbol{\mu}) = [v_{ij}(\boldsymbol{\mu})]$

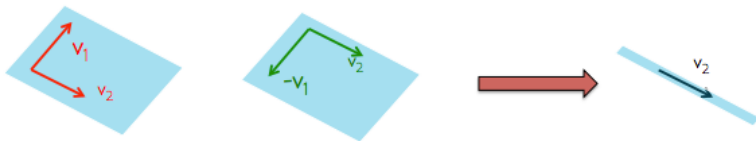
Direct interpolation doesn't work

- Example

- $N_{\mathbf{w}} = 3, k = 2, p = 1$
- for $\mu_1 = 0, \mathbf{V}(0) = [\mathbf{v}_1, \mathbf{v}_2]$
- for $\mu_2 = 1, \mathbf{V}(1) = [-\mathbf{v}_1, \mathbf{v}_2]$
- target parameter $\mu = 0.5$
- use linear interpolation

- Interpolation result:

$$\mathbf{V}(0.5) = 0.5(\mathbf{V}(0) + \mathbf{V}(1)) = [0.5(\mathbf{v}_1 - \mathbf{v}_1), 0.5(\mathbf{v}_2 + \mathbf{v}_2)] = [0, \mathbf{v}_2]$$



- What went wrong?
- We haven't interpolated the correct object

Subspace interpolation

- Projection-based model reduction

$$\frac{d\mathbf{q}}{dt}(t) = \mathbf{V}(\boldsymbol{\mu})^T \mathbf{A}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}) \mathbf{q}(t) + \mathbf{V}(\boldsymbol{\mu})^T \mathbf{B} \mathbf{u}(t)$$

- Equivalent full state equation (multiply by $\mathbf{V}(\boldsymbol{\mu})$)

$$\frac{d\mathbf{w}}{dt}(t) = \boldsymbol{\Pi}_{\mathbf{V}(\boldsymbol{\mu}), \mathbf{V}(\boldsymbol{\mu})} \mathbf{A}(\boldsymbol{\mu}) \mathbf{w}(t) + \boldsymbol{\Pi}_{\mathbf{V}(\boldsymbol{\mu}), \mathbf{V}(\boldsymbol{\mu})} \mathbf{B} \mathbf{u}(t)$$

- An orthogonal projection is independent of the choice of reduced basis associated to the projection subspace
- Important quantity to interpolate: subspace

The Grassmann manifold

- A subspace \mathcal{S} is typically represented by a reduced basis
- The choice of reduced basis is not unique

$$\mathcal{S} = \text{range}(\mathbf{V}) = \text{range}(\mathbf{V}\mathbf{Q}), \quad \forall \mathbf{Q} \in \text{GL}(k)$$

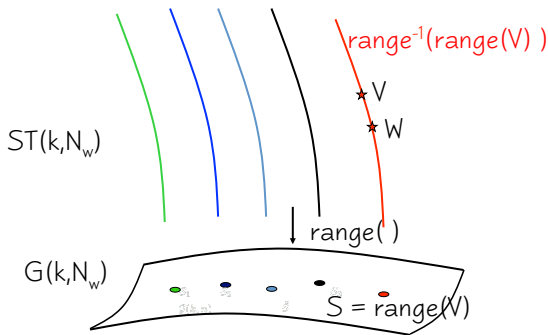
- Matrix manifolds of interest
 - $\mathcal{G}(k, N_{\mathbf{w}})$ (Grassmann manifold): set of subspaces of dimension k in $\mathbb{R}^{N_{\mathbf{w}}}$
 - $\mathcal{ST}(k, N_{\mathbf{w}})$ (orthogonal Stiefel manifold): set of orthogonal reduced bases of dimension k in $\mathbb{R}^{N_{\mathbf{w}}}$
- Case of model reduction
 - $\mathbf{V}(\boldsymbol{\mu}) \in \mathcal{ST}(k, N_{\mathbf{w}})$
 - $\text{range}(\mathbf{V}(\boldsymbol{\mu})) \in \mathcal{G}(k, N_{\mathbf{w}})$
- Interpolation on the Grassmann manifold (quotient manifold) using quantities belonging to the Stiefel manifold

$$\mathcal{G}(k, N_{\mathbf{w}}) = \mathcal{ST}(k, N_{\mathbf{w}}) / \mathcal{O}(k)$$

The Grassmann manifold

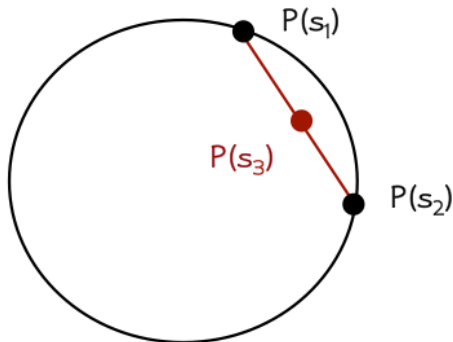
- Matrix manifolds of interest

- $\mathcal{G}(k, N_w)$ (Grassmann manifold): set of subspaces of dimension k in \mathbb{R}^{N_w}
- $\mathcal{ST}(k, N_w)$ (orthogonal Stiefel manifold): set of orthogonal reduced bases of dimension k in \mathbb{R}^{N_w}



Interpolation on matrix manifolds

- First example: the circle



- Standard interpolation fails
- Idea: interpolate on a linear space \Rightarrow a tangent space to the manifold

Interpolation on matrix manifolds

- Input:
 - precomputed matrices $\{\mathbf{A}(\boldsymbol{\mu}_l)\}_{l=1}^m$
 - multi-variate interpolation operator $a(\boldsymbol{\mu}) = \mathcal{I}(\boldsymbol{\mu}; \{a(\boldsymbol{\mu}_l), \boldsymbol{\mu}_l\}_{l=1}^m)$
 - map $m_{\mathbf{A}}$ from the manifold \mathcal{M} to the tangent space of \mathcal{M} at $m_{\mathbf{A}}$
 - inverse map $m_{\mathbf{A}}^{-1}$ from the tangent space to \mathcal{M} at $m_{\mathbf{A}}$ to the manifold \mathcal{M}

```
1: for  $l = 1 : m$  do  
2:   Compute  $\Gamma(\boldsymbol{\mu}_l) = m_{\mathbf{A}}(\mathbf{A}(\boldsymbol{\mu}_l))$   
3: end for  
4: for  $i = 1 : N_{\mathbf{w}}$  do  
5:   for  $j = 1 : k$  do  
6:     Compute  $\Gamma_{ij}(\boldsymbol{\mu}_l) = \mathcal{I}(\boldsymbol{\mu}; \{\Gamma_{ij}(\boldsymbol{\mu}_l), \boldsymbol{\mu}_l\}_{l=1}^m)$   
7:   end for  
8: end for  
9: Form  $\Gamma(\boldsymbol{\mu}) = [\Gamma_{ij}(\boldsymbol{\mu})]$  and compute  $\mathbf{A}(\boldsymbol{\mu}) = m_{\mathbf{A}}^{-1}(\Gamma(\boldsymbol{\mu}))$ 
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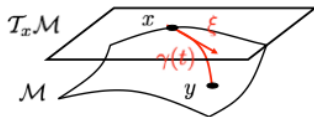
- Requirement: the interpolation operator \mathcal{I} preserves the tangent space.

for instance:
$$a(\boldsymbol{\mu}) = \mathcal{I}(\boldsymbol{\mu}; \{a(\boldsymbol{\mu}_l), \boldsymbol{\mu}_l\}_{l=1}^m) = \sum_{l=1}^m \theta_l(\boldsymbol{\mu}) a(\boldsymbol{\mu}_l)$$

Interpolation on matrix manifolds

- How do we find $m_{\mathbf{A}}$ and its inverse $m_{\mathbf{A}}^{-1}$
- Idea: use concepts from differential geometry
- Geodesics
 - generalize straight lines on manifolds
 - uniquely defined given a point x of the manifold and a tangent vector ξ at this point

$$\begin{aligned}\gamma(t; x, \xi) &: [0, 1] \rightarrow \mathcal{M} \\ \gamma(0; x, \xi) &= x, \quad \dot{\gamma}(0, x, \xi) = \xi\end{aligned}$$



- Exponential map

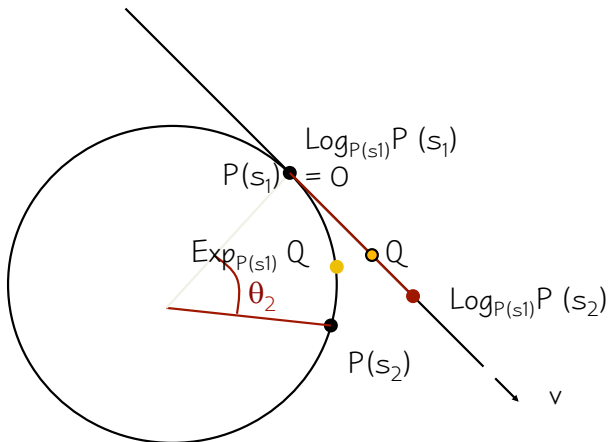
$$\text{Exp}_x : \mathcal{T}_x \mathcal{M} \rightarrow \mathcal{M} \quad \xi \mapsto \gamma(1; x, \xi)$$

- Logarithm map (defined in a neighborhood \mathcal{U}_x of x)

$$\text{Log}_x : \mathcal{U}_x \subset \mathcal{M} \rightarrow \mathcal{T}_x \mathcal{M} \quad y \mapsto \text{Exp}_x^{-1}(y)$$

Interpolation on matrix manifolds

- Application to interpolation of points on a circle



Interpolation on matrix manifolds

Case of the Grassmann manifold

- Logarithmic map
 - 1 Compute the thin SVD

$$(\mathbf{I} - \mathbf{V}_0 \mathbf{V}_0^T) \mathbf{V}_1 (\mathbf{V}_0^T \mathbf{V}_1)^{-1} = \mathbf{U} \mathbf{\Sigma} \mathbf{Z}^T$$

- 2 Compute

$$\mathbf{\Gamma} = \mathbf{U} \tan^{-1}(\mathbf{\Sigma}) \mathbf{Z}^T$$

- 3 $\text{Log}_{\mathcal{S}_0}(\mathcal{S}_1) \leftrightarrow \mathbf{\Gamma}$

- Exponential map of $\xi \in \mathcal{T}_{\mathcal{S}_0} \mathcal{G} \leftrightarrow \mathbf{\Gamma}$

- 1 Compute the thin SVD

$$\mathbf{\Gamma} = \mathbf{U} \mathbf{\Sigma} \mathbf{Z}^T$$

- 2 Compute

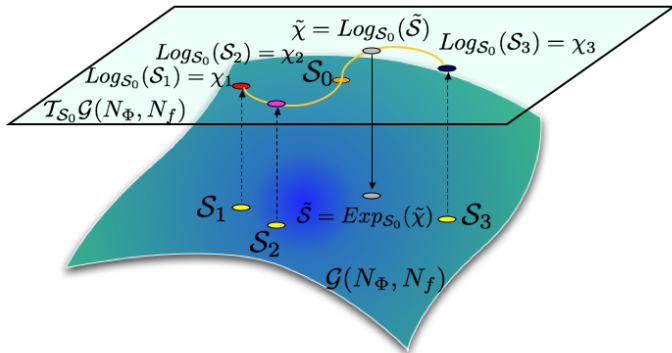
$$\mathbf{V} = \mathbf{V}_0 \mathbf{Z} \cos \mathbf{\Sigma} + \mathbf{U} \sin \mathbf{\Sigma}$$

- 3 $\text{Exp}_{\mathcal{S}_0}(\xi) = \text{range}(\mathbf{V})$

- Note: the trigonometric operators only apply to the diagonal elements of the matrices

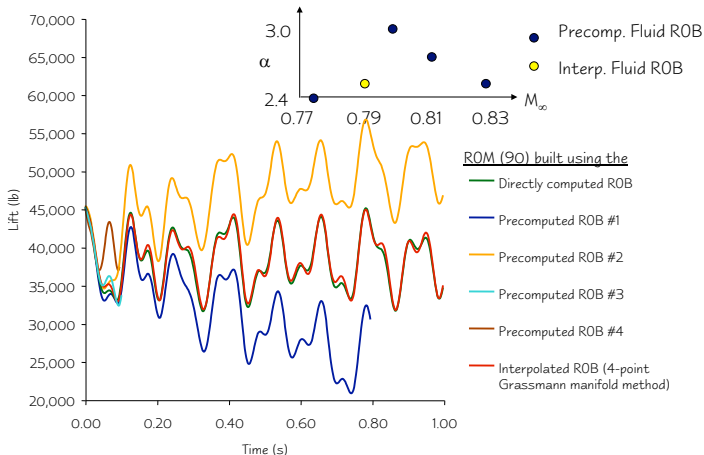
Interpolation on matrix manifolds

- Case of the Grassmann manifold



Application to aeroelasticity

- Aeroelastic behavior of the F-16



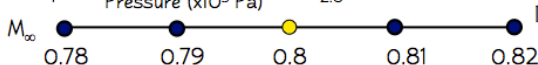
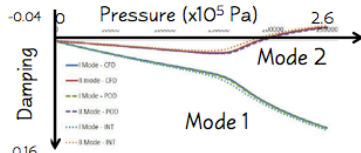
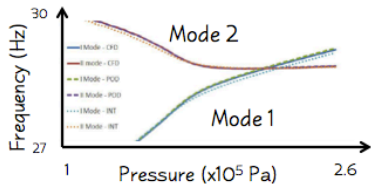
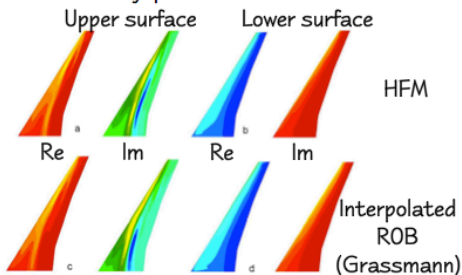
Application to aeroelasticity

- Aeroelastic behavior a commercial aircraft

Airbus AMP model



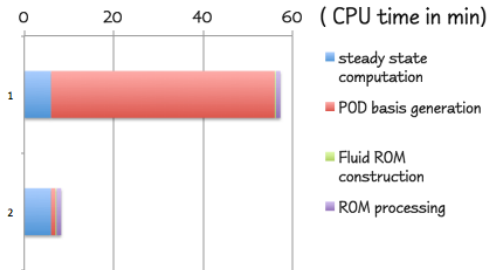
Unsteady pressure distribution



ROM (46) Vetrano et al.,
ASD Journal 2011

Application to aeroelasticity

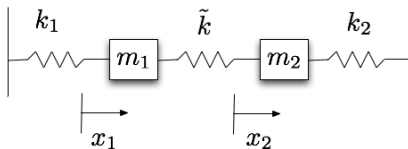
- Aeroelastic behavior of the F-16



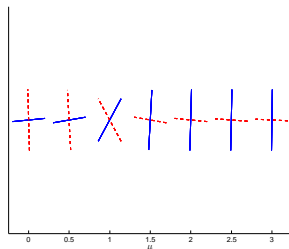
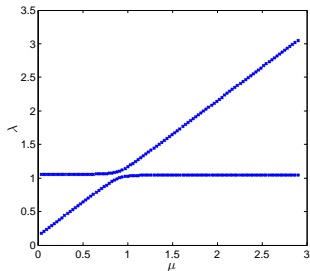
- Dominant cost: computation of $\mathbf{A}(\mu)$ and $\mathbf{B}(\mu)$ for a new value of μ
- Approach #2: interpolate $(\mathbf{V}(\mu)^T \mathbf{A}(\mu) \mathbf{V}(\mu), \mathbf{V}(\mu)^T \mathbf{B}(\mu))$

Example

- Simple example: mass-spring system with two degrees of freedom
- $\mu = k_1 - 0.1$

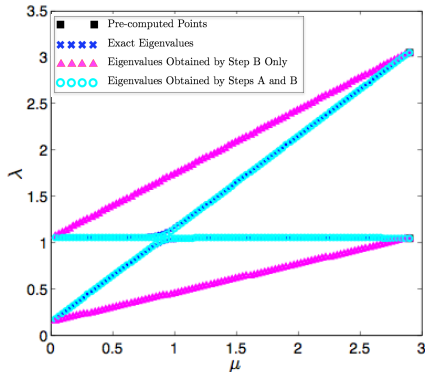


- $\mathbf{V}(\mu)^T \mathbf{A}(\mu) \mathbf{V}(\mu) = \mathbf{\Lambda}(\mu)$



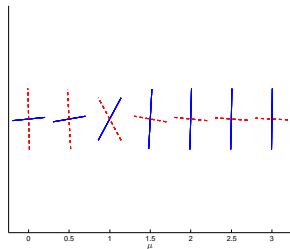
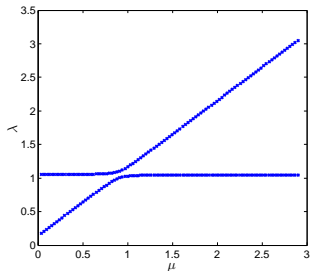
Interpolation on a matrix manifold

- $\Lambda(\mu)$ belongs to the manifold of symmetric positive definite matrices (diagonal)
- Interpolate on the manifold using $(\Lambda(0), \Lambda(2.9))$



Example: Mode veering and mode crossing

- The issue is the mode veering: the coordinates of the reduced matrices are not consistent



- There would be an issue also with mode crossing (the eigenfrequencies are ordered increasingly in Λ)

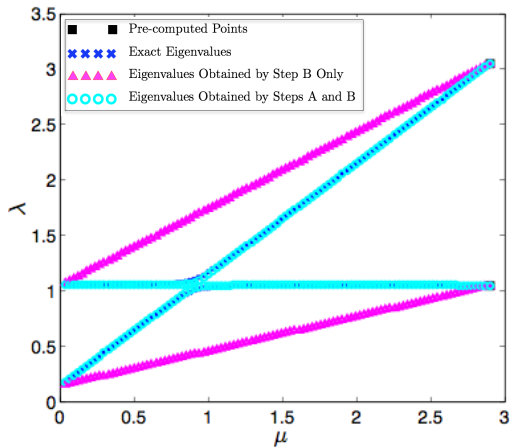
Consistent interpolation on matrix manifolds

- Solution: pre-process the reduced matrices (Step A)
- Consistency enforced by the solution of an orthogonal Procrustes problem

$$\min_{\mathbf{Q}_i} \|\mathbf{V}_i \mathbf{Q}_i - \mathbf{V}_{i_0}\|_F, \quad \forall i = 1, \dots, m$$

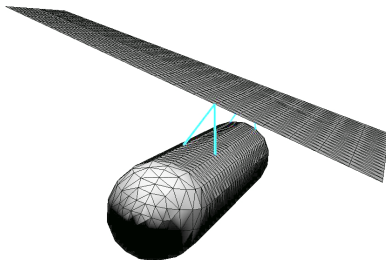
- Analytical solution using the SVD
 - 1 Compute $\mathbf{P}_{i,i_0} = \mathbf{V}_i^T \mathbf{V}_{i_0}$
 - 2 Compute the SVD $\mathbf{P}_{i,i_0} = \mathbf{U}_{i,i_0} \mathbf{\Sigma}_{i,i_0} \mathbf{Z}_{i,i_0}^T$
 - 3 Let $\mathbf{Q}_i = \mathbf{U}_{i,i_0} \mathbf{Z}_{i,i_0}^T$
- Can be processed online or offline
- Step B: interpolation on a matrix manifold

Application 1

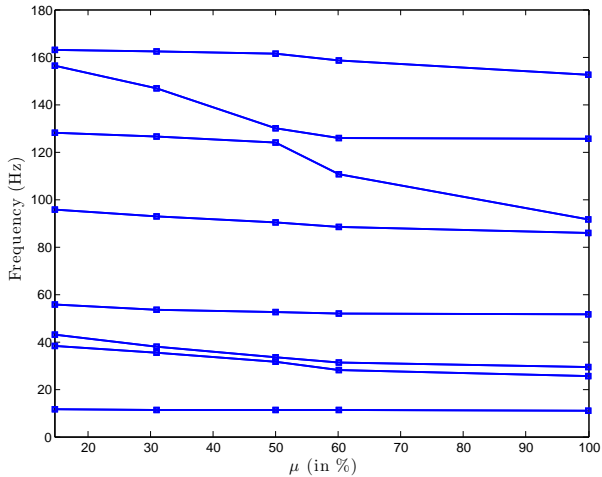


Application 2

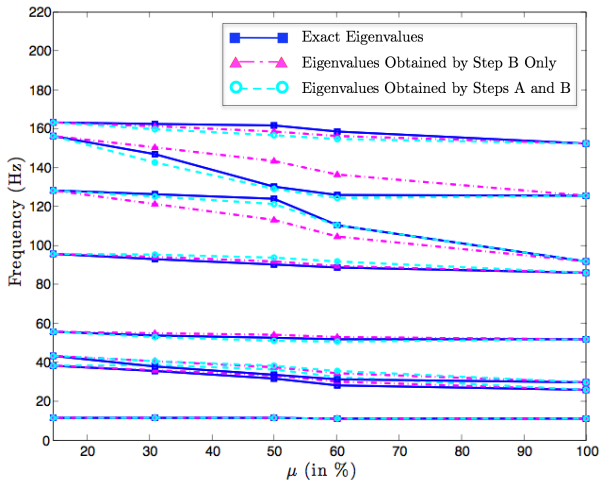
- More challenging example: wing with tank and sloshing effect
- The hydro-elastic effects affect the eigen-frequencies and eigen-modes of the structure
- The parameter μ defines the level of fuel in the tank $0 \leq \mu \leq 100\%$



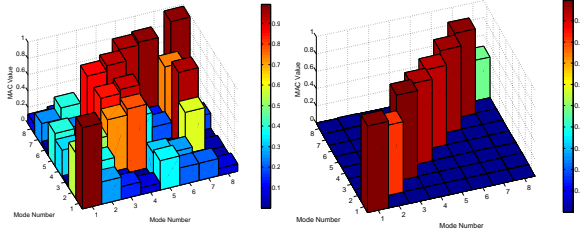
Application 2



Application 2



Link with Modal Assurance Criterion



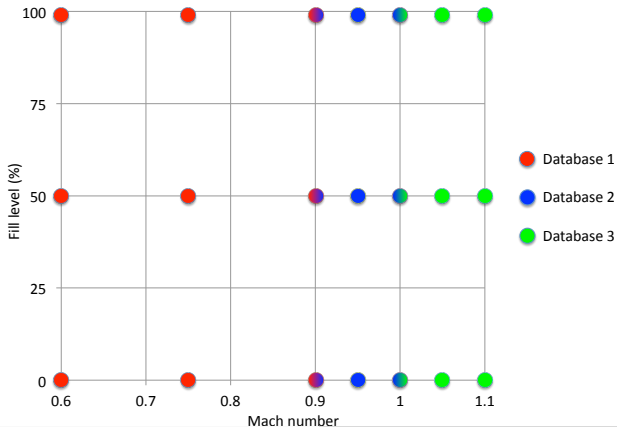
- The MAC between two eigenmodes ϕ and ψ is

$$\text{MAC}(\phi, \psi) = \frac{|\phi^T \psi|^2}{(\phi^T \phi)(\psi^T \psi)}$$

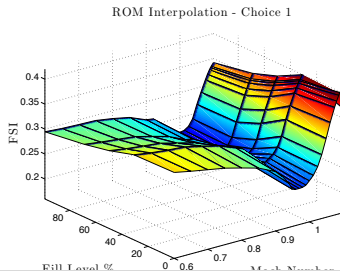
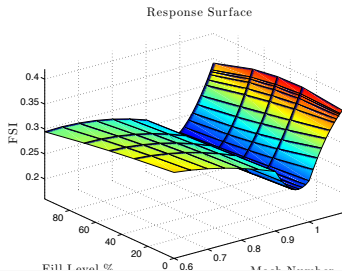
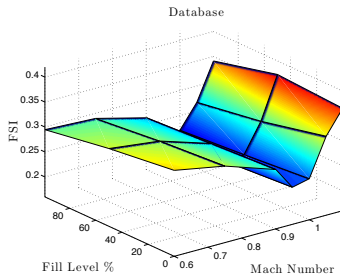
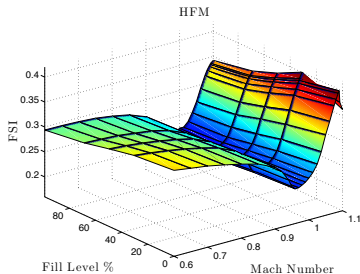
- When ϕ and ψ are normalized $\text{MAC}(\phi, \psi) = |\phi^T \psi|^2$
- \mathbf{P}_{i,i_0} is the matrix of square roots of the MACs between the modes contained in $\mathbf{V}(\mu_i)$ and those contained in $\mathbf{V}(\mu_{i_0})$.
- This is the Modal Assurance Criterion Square Root (MACSR)

Application 3

- Aeroelastic study of the wing-tank system
- 2 parameters: fill level and free-stream Mach number M_∞
- Database approach



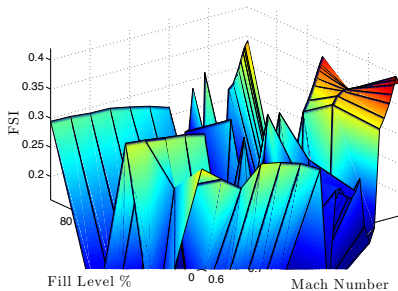
Application 3



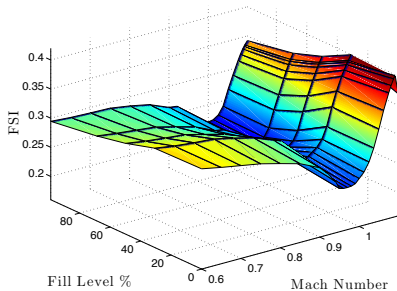
Application 3

- Effect of Step A

ROM Interpolation - Choice 1

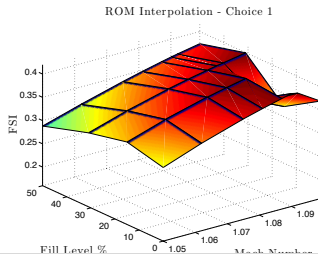
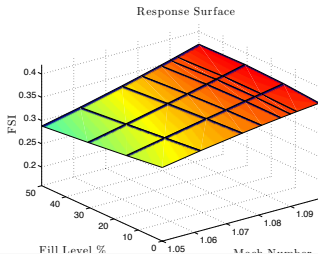
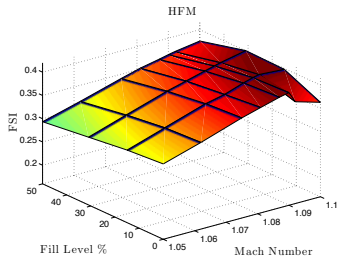


ROM Interpolation - Choice 1

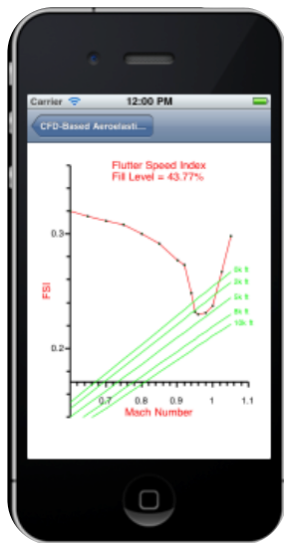


Application 3

- Bifurcation detection



Mobile computing using a database of ROMs



References

- D. Amsallem, C. Farhat: Interpolation method for adapting reduced-order models and application to aeroelasticity. *AIAA Journal* 46(7), 1803–1813 (2008)
- D. Amsallem, C. Farhat: An online method for interpolating linear parametric reduced- order models. *SIAM Journal on Scientific Computing* 33(5), 2169–2198 (2011)
- D. Amsallem, Interpolation on Manifolds of CFD-based Fluid and Finite Element-based Structural Reduced-order Models for On-line Aeroelastic Prediction, PhD thesis, Stanford University (2010)